

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10014090 (EASA.A.S.01187)	ROEDER PRAEZISION GMBH	REV. 1		SA227-BC, SA227-DC, SA227-TT SA226-TC, SA227-AC, SA227-AT, SA226-AT, SA226-T(B), SA226-T,	Installation of Thommen Air Data Display and Altimeter.	The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.	Flight Manual Supplement Doc No SA226/227-NY-0212-08, Rev. 0, or later EASA approved revision.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.	25/09/2017
10014142 (EASA.A.S.01256)	ROEDER PRAEZISION GMBH	REV. 2		A330-322	Cabin reconfiguration from a 18 B/C + 331 Y/C layout to a 18 B/C +343Y/C layout. This STC installs a new cabin arrangement.	The Certification Basis (CB) for the original product remains applicable to this certificate/approval. As per EASA TCDS A.004 Elect to Comply with CS25 (Original Issue) requirements: 25.0811(d)/ 25.812(d)(e)/ 25.813(a)(b)(e)/ 25.851(a)/25.853(a)(c)(f)(h)/ 25.869(a)(b)(c). The requirements for environment...	Master Document List, ref. MDL330-25-01-05 Revision 2, dated 27 Apr. 2005 (>,<)> or later revisions of the above listed documents approved by EASA.	Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.	15/06/2018
10014426 (EASA.A.S.02471)	ROEDER PRAEZISION GMBH	REV. 1		441	Installation of Dual Mode S Transponder GTX 330D	The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.	- Engineering Order C441-NY-0124- AFM-S C441-NY-0124-08, Rev. 0, approved 14. February 2006, or later EASA approved revision- Maintenance Procedures C441-NY-0124-07, Rev. 0, or later approved revision.	1. Limitation according AFM-S C441-NY-0124-082. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...	25/09/2017
10014439 (EASA.A.S.02489)	ROEDER PRAEZISION GMBH	REV. 2		HAWKER 800/-800XP/-850XP BAE.125 SERIES 800A/-800B BAE.125 SERIES 1000A/-1000B	Installation of wing mounted Xenon Taxi/Landing Light and/or Radom mounted Taxi Light	The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Engineering Order H800-NY-0125-05- AFM-S H800-NY-0125-08, Rev 0 approved 07/04/2006- Maintenance Procedures H800-NY-0125-07, Rev 0 or later revisions of the above listed documents approved by EASA.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.	13/02/2018

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10014440 (EASA.A.S.02490)	ROEDER PRAEZISION GMBH	REV. 1		HAWKER 800/-800XP BAE.125 SERIES 800A/-800B BAE.125 SERIES 1000A/- 1000B	Installation of Iridium Satcom C1000	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Engineering Order H800-NY-0126-01- AFM-S H800-NY-0126-08, Rev. 0, approved 12. May 2006- Maintenance Procedures H800-NY-0125-07, Rev. 0 or later revisions of the above listed documents approved by EASA.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	13/02/2018
10014527 (EASA.A.S.02612)	ROEDER PRAEZISION GMBH	REV. 1		CL-600-2D24 (RJ SERIES 900) CL-600-2C10 (RJ 700,701,702) CL-600-2B19 (RJ SERIES 100)	Installation of RAAS compatible EGPWS Computer in CL-600- 2B19/-2C10/-2D24 Series Aircraft	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Document Inventory List CRJ-NY-0429 Revision 1, dated 09.02.2007- Flight Manual Supplement CRJ-NY-0429-08 Revision 0 or later revisions of the above listed documents approved by EASA.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	13/02/2018
10014528 (EASA.A.S.02613)	ROEDER PRAEZISION GMBH	REV. 1		AVRO 146-RJ85 AVRO 146-RJ100	Activation of RAAS function of EGPWS computer in AVRO- RJ85/AVRO- RJ100 Series Aircraft	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	AFM - RJ85-NY-0428-08 Rev 1 Master Document List RJ85-NY-0428-Rev 0	1. As listed in the AFM. 2. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	25/09/2017
10014537 (EASA.A.S.02624)	ROEDER PRAEZISION GMBH	REV. 2		RJ85/100	Installation of RAAS compatible EGPWS Computer in AVRO RJ85/100 Series Aircraft.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- AFM - RJ85-NY-0427-08 Rev. 0- Master Document List RJ85-NY-0427-08 Rev. 0 or later revisions of the above listed documents approved by EASA.	- As listed in the AFM.- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	10/10/2017

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10014538 (EASA.A.S.02625)	ROEDER PRAEZISION GMBH	REV. 1		CL-600-2D24 (RJ SERIES 900) CL-600-2C10 (RJ 700,701,702) CL-600-2B19 (RJ SERIES 100)	Activation of RAAS function of EGPWS Computer in CL-600-2B19/-2C10/-2D24 Series Aircraft	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Document Inventory List CRJ-NY-0432 Revision 1, dated 09.02.2007- Flight Manual Supplement CRJ-NY-0432-08 Revision 0 or later revisions of the above listed documents approved by EASA.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	13/02/2018
10014604 (EASA.A.S.02727)	ROEDER PRAEZISION GMBH	REV. 2		DC-3C-R1830-90C	Installation of emergency exit signs and floor escape path markings at DC3C series aircraft Installation of emergency exit signs light with power source and installation of fluorescent floor path markings	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Master Document List DC3-NY-0532 Rev. 0 dated 20 Nov 2006 or later revisions of the above listed documents approved by EASA.- Engineering Order DC3-NY-0532-05 Rev.0- Mechanical Installation DC3-NY-0532-03 Rev. 0- Electrical Drawings DC3-NY-0531-02 Rev. 0- Weight & Balance DC3-NY-0532-09 Rev. 0	The system must be maintained in accordance with Maintenance Procedures DC3-NY-0532-07 Rev. 0 dated 20 Nov 2006 or later approved revision. The system must be operated in accordance with AFM supplement DC3-NY-0532-08 Rev. 0 dated 20 Nov 2006 or later approved revision. Prior to installation of this desi...	08/10/2018
10014612 (EASA.A.S.02739)	ROEDER PRAEZISION GMBH	REV. 1		200, 300, 1900	Installation of HID Nose Gear Lights.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Installation to be in accordance with Nayak Engineering Order ref BE200/300-NY-0733-05, Rev 0 dated 12 September 2006, or later approved revision.- Instructions for Continued Airworthiness as defined in Nayak Maintenance Procedures ref BE200/300-NY-0733-07, Rev 0 dated 12 September 2006, or later appr...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	30/10/2017

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10014676 (EASA.A.S.02837)	ROEDER PRAEZISION GMBH	REV. 1		737-300	Installation of Galley G2 Varig p/n 25-30-3154 inside aircraft between STA 333.25 and STA 350.50	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Documentation according to "Uebersicht der Dokumente" Rev. 3 dated 30.01.07- Compliance Check List B737-NY-0936-01A, Rev. 2 dated 30.01.07- Declaration of Completion B737-NY-0936, dated 30.01.07 or later revisions of the above listed documents approved by EASA.	This STC is limited to Boeing 737-300 MSN 23827. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product...	13/02/2018
10014753 (EASA.A.S.02939)	ROEDER PRAEZISION GMBH	REV. 1		DC-9-87 (MD-87) DC-9-83 (MD-83) DC-9-82 (MD-82) DC-9-81 (MD-81)	Installation of flight ID capable XPDR and CTRL		Ubersicht der Dokumente No. MD80-NY-1137 dated 14.02.2007 AFMS No. MD80-NY-1137-08 dated 14.02.2007 or later revisions of the above listed documents approved by EASA.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017
10014804 (EASA.A.S.03007)	ROEDER PRAEZISION GMBH	REV. 1		425	Installation of GTX 330D Mode S Transponder	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Doc. No. EO C425-NY-1240-05, R0.- EASA approved NAYAK AFM-S Doc. No. C425-NY-1240-08, or later approved revision.- NAYAK Maintenance Procedures. Doc. No. C425-NY-1240-07, or later approved revision.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	30/10/2017

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10014881 (EASA.A.S.03109)	ROEDER PRAEZISION GMBH	REV. 1		MU-2B-40, MU-2B-60	Installation of XPDR Mode-S System with Diversity, Garmin GTX 330 D (single or optional dual installation possible).	The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.	- EASA approved Compliance Check List / Compliance Statement, Doc. No. MU2-NY-0243-01A, Revision 0, dated 20 February 2007 or later approved revision. - EASA approved Airplane Flight Manual Supplement, Doc. No. MU2-NY-0243-08, dated 20-February 2007, or later approved revision. - Approved Amendment to Air...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.	30/10/2017
10014924 (EASA.A.S.03172)	ROEDER PRAEZISION GMBH	REV. 2		A330-322	Business Class extension to a 30 B/C + 295 Y/C layout. This STC modifies cabin arrangement from a 18 B/C + 343 Y/C layout to a 30 B/C + 295 Y/C layout with new B/C seat and stowage compartment installations.	The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS25 Amdt 3. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original p...	Master Document List, ref. MDL330-25-01-07 Revision 1, dated 25 Oct. 2007 (>,<) or later revisions of the above listed documents approved by EASA.	Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.	15/06/2018
10014971 (EASA.A.S.03251)	ROEDER PRAEZISION GMBH	REV. 1		SA227-AT, SA227-TT, SA227-DC SA226-T, SA226-AT, SA226-T(B),	Installation of Honeywell MK IV EGPWS, together with MFD and Thommen Air Data Computer	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	EASA approved „Übersicht der Dokumente“, Project No. SA226/227-NY-0808, Revision 0, dated 07-Nov-2007, or later approved revision EASA approved Airplane Flight Manual Supplement, Doc. No. SA226/227-NY-0808-08, Revision 0, dated 31-May-2007, or later approved revision Maintenance Procedures, Doc. No. SA22...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017

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10015257 (EASA.A.S.03710)	ROEDER PRAEZISION GMBH	REV. 1		PA-34-200 /-200T/-220T	Replacement of existing Garmin COM/NAV and existing Garmin GPS system by Garmin GNS 530w COM/NAV/GPS system.	The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.	- Approved Engineering Order, Doc. No. PA34-NY-10118-05, Revision 1, dated 26 January 2009, or later approved revision. -EASA approved Airplane Flight Manual Supplement, Doc. No. PA34-NY-10118-08, Revision 2, dated 27 January 2009 or later approved revision. - Approved Maintenance Procedure, Doc. No. PA...	1. Limited to "Single GNS" – Installation only. 2. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...	30/10/2017
10015303 (EASA.A.S.03868)	ROEDER PRAEZISION GMBH	REV. 1		A319-112	Change of the Cabin Configuration (150 Pax). The cabin configuration will be changed to increase the certified 145 pax configuration to a 150 pax configuration with minimum Cabin Crew of 3 Flight Attendants.	The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Certification Report, ref. doc. A319-CR25-03-2009, Issue 1, dated 17th March 2009. - Design Modification Document, ref. doc. A319-DMD25-03-2009, Issue 1, dated 17th March 2009. - Compliance Record Document, Issue 2, dated 5th May 2009.	Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is s...	15/06/2018
10028781	ROEDER PRAEZISION GMBH	REV. 1		210	Post Installation Approval and conformity check of avionic system.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	-Flight Manual Supplement: C210-NY-07136-08, Rev. 1.- Instructions for Continued Airworthiness: C210-NY-07136-07, Rev. 0.-Installation Manual: C210-NY-07136-05, Rev. 1 or later revisions of the above listed documents approved by EASA.	Limited to S.N 57339. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017

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10029201	ROEDER PRAEZISION GMBH			CESSNA P210N	Ferry Tank Provisioning	The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	-Engineering Order 725EO280010-01A-"POH Supplement for Cessna P210N, S/N P21000629", Doc. No. 7725 AM 280010 01A, Revision 1, dated 18 February 2010 or later revisions of the above listed documents approved by EASA	Limited to Aircraft S/N P210 00629	11/03/2010
10030066	ROEDER PRAEZISION GMBH	REV. 2		ATR 72-212, ATR 72-212A ATR 72-202, ATR 72-211 ATR 72-102, ATR 72-201 ATR 42-500, ATR 72-101 ATR 42-300, ATR 42-320	Installation of single HF-9000 Communication Transceiver System in ATR42 and ATR 72. This STC installs an additional HF communication device. This revision has added some ATR 42 aircraft models within the initial STC applicability.	The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): H-01: Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWISCS 25 Amdt 4:25.21, .25, .27, .301, .303 to .307, .561, .571, .581, 25.601 t...	-Master Document List, ref: ATR72-NY-02155 Revision 7 dated 16 April 2013-Aircraft Flight Manual Supplement, ref: ATR72-NY-02155-08 Revision 5 dated 09 August 2012-Instructions for Continued Airworthiness, ref: ATR72-NY-02155-07 Revision 6 dated 20 November 2012-Weight & Balance Supplement, ref: ATR72-...	ATR Modification 2028 (Fuselage- Provide basic provisions for HF antenna strap insulator) and Mod ATR Modification 3138 (Communications-HF system installation of basic antenna) must be embodied on the aircraft at the time of this STC installation. Prior to installation of this design change it must be determined...	15/11/2017
10030847	ROEDER PRAEZISION GMBH	REV. 1		206H, T206H	Provisioning for implementation and operation with a "Turtle Pac160-ferry tank"	The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23, Initial Issue (not required by 21A.101, elected to comply by the applicant). The requirements for environmental protection and the ass...	-Engineering Order 1704EO280010-01A, dated 26.06.2010-POH Supplement: "Anhang zum Flughandbuch für Cessna 206H and T206H", Doc. No. 1704 AM 2800 10 01A, dated 18.05.2010 or later revisions of the above listed documents approved by EASA	Installation of the ferry tank is not subject of this STC.	14/07/2010

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10031264	ROEDER PRAEZISION GMBH	REV. 1		C90A, C90GT, C90GTI, C90	Installation of an Avidyne Traffic Advisory System TAS 600/610/620	The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain ap...	Master Document List, Project No. C90-NY-07137, Rev. 1, dated 03 August 2010, or later approved revision. Master Document List, Project No. C90-NY-07137, Rev. 1, dated 3 August 2010, or later approved revision. Flight Manual Supplement, Doc. No. C90-NY-07137-08, Rev. 1, dated 2 August 2010, or later appr...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017
10033154	ROEDER PRAEZISION GMBH	REV. 1		SA26-T, SA26-AT, SA226-T SA227-TT, SA227-AC, SA227-DC SA227-AT SA226-AT, SA226-T(B), SA226-TC	Dual Installation of GNS 430 WAAS in SA226/227 Series	The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unc...	Instructions for Installation: GNS430-NY-03157-05, Rev. 2, dated 05.08.2010 Flight Manual Supplements: GNS430-NY-03157-08, Rev. 3, dated 05.08.2010 Instructions for Continued Airworthiness: GNS430-NY-03157-07, Rev. 1, dated 05.08.2010 or later revisions of the above listed documents approved by EASA	Installation of 3rd dissimilar and independent COM system required to IFR operation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon...	15/11/2017
10033946	ROEDER PRAEZISION GMBH	REV. 1		421A/421B/421C/425 402C/411/411A/414/414A/421 401/401A/401B/402/402A/402B	Installation of a Traffic Advisory System Avidyne TAS610/620.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Engineering Order C421-NY-08167-05, Rev. 1. Maintenance Procedures C421-NY-08167-07, Rev. 1. Flight Manual Supplement C421-NY-08167-07, Rev. 1 or later revisions of the above listed documents approved by EASA.	A compatible Multi Function Display (MFD) must have been already installed. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airwor...	15/11/2017



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10033970	ROEDER PRAEZISION GMBH	REV. 3		421A/421B/421C/425 402C/411/411A/414/414A/ 421 401/401A/401B/402/402A/ 402B	Single installation of Garmin GNS 430 (A)W/GNS 530 (A)W and optional installation of Roll Steering Converter GDC31.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Engineering Order C421-NY-08168-05, Rev. 2. Flight Manual Supplement: C421-NY-08168-08a, Rev. 2; C421-NY-08168-08b, Rev. 1. Maintenance Procedures: C421-NY-08168-07, Rev. 1 or later revisions of the above listed documents approved by EASA.	An independent COMM or NAV/DME or ADF/DME must have been already installed. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airwor...	15/11/2017
10034119	ROEDER PRAEZISION GMBH	REV. 1		PIPER PA-31-350 (CHIEFTAIN)	Provision for Camera Installation.	The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Installation in accordance with Master Record List Document No. 71335MR 5300 10 01B- Operation in accordance with Supplement to Pilot Operating Handbook POH, Document No. 71335 AM 5300 10 01B- Maintenance in accordance with Instructions for Continued Airworthiness No. 71335 CA 5300 10 01B or later re...	This STC revision includes only the structural changes. It does not include any installation of equipment (e.g. Laser scanner, computer, cameras) Prior to installation of this change/ repair it must be determined that the interrelationship between this change/ repair and any other previously installed chan...	10/08/2018
10034483	ROEDER PRAEZISION GMBH	REV. 1		560XL	Installation of HID/LED LDG/REC/TAXI light. Replacement of existing conventional LDG/REC/TAXI lights by low power HID/LED lights.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Master Document List to project C560XL-NY-1170, Rev. 3 from 07 April 2011. ICA C560XL-NY-1170-07, Rev. 1 from 07 April 2011 or later revisions of the above listed documents approved by EASA.	Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10034925	ROEDER PRAEZISION GMBH	REV. 2		T303	Installation of 1. Dual GPS/NAV/COMM Garmin GNS430/530(A)W2.E-HSI Sandel SN3308 with Directional Gyro Century NSD360A3. Various other NAV/COMM related Avionic Components	The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	-Engineering Order: C303-NY-07165-05, Rev. 1-Flight Manual Supplements: C303-NY-07165-08a, Rev. 1C303-NY-07165-08b, Rev. 1C303-NY-07165-08c, Rev. 1C303-NY-07165-08d, Rev. 1C303-NY-07165-08e, Rev. 1C303-NY-07165-08f, Rev. 1- Maintenance Procedures: C303-NY-07165-07, Rev. 1 or later revisions of the above...	This STC consists of three blocks (see title) which may be installed independently from each other. The components of the third block (see Equipment List) may also be installed independently. The third block of this STC includes the installation of: Honeywell (Bendix/King) ADF KR-87 with indicator KI227...	15/11/2017
10036680	ROEDER PRAEZISION GMBH	REV. 1		PA-46R-350T PA-46-310P, -350P, -500TP,	Installation of Tactical Weather Detection System (Stormscope) Avidyne TWX670.	The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original...	Flight Manual-Supplements: TWX670-NY-02180-08, Rev. 1 Engineering Order: TWX670-NY-02180-05, Rev. 0 Maintenance Procedures: TWX670-NY-02180-07, Rev. 0 or later revisions of the above listed documents approved by EASA.	This STC is also applicable to Turbine Converted Aircraft (PA46-310-DLX, PA46-350DLX) according to FAA STC ST00541SE. The aircraft must have been already equipped with a suitable Multifunction Display (MFD). Prior to installation of this modification it must be determined that the interrelationship between...	15/11/2017
10036936	ROEDER PRAEZISION GMBH	REV. 1		F27 MK050	Installation of cabin divider	The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Cabin divider installation according to EO F50-NY-12177-05 Incorporation of maintenance instructions according to EO F50-NY-12177-07	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10037159	ROEDER PRAEZISION GMBH	REV. 1		A320-214	A320 Cabin Reconfiguration for AZAL.	The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval JAR 25 Change 11 and EASA CS 25.853 Initial Issue, as per Air Berlin Technik's Certification Report A320-CR25-06-2011 Revision 1. In addition...	- Master Documentation List A320-MDL25-06-2011 Revision Initial dated 05 November 2011; and - Engineering Order A320-EO25-06-2011 Revision 1 dated 21 October 2011; or later revisions of the above listed documents approved by EASA.	- This design change (STC) is applicable to the following Airbus A320-214 serial number aircraft (MSNs): 2685; 2846; 2853; 2865; 2991 and 3006. This STC does not require a revision or supplement to the original TCHolder's EWIS Instructions for Continued Airworthiness. Prior to installation of this modification...	15/06/2018
10037575	ROEDER PRAEZISION GMBH	REV. 2		EMB-145MK, EMB-145EU EMB-145LU EMB-145ER, EMB-145LR EMB-145EP, EMB-145MP EMB-135ER, EMB-135LR EMB-135BJ	Installation of HID Taxi/Landing light, inspection light on EMB-145 aircraft.	The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Master Document List doc. N° EMB145-NY-04188-MDL Rev. 2 dated 07.12.2011. Airplane Flight Manual Supplement doc. N° EMB145-NY-04188-08 Rev. 4 dated 02.12.2011. Instructions for Continued Airworthiness - Maintenance procedures doc. N° EMB145-NY-04188-07 Rev. 3 dated 02.12.2011 or later revisions of the above...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017
10038427	ROEDER PRAEZISION GMBH	REV. 2		PA-46-500TP, PA-46R350T PA-46-310P, PA-46-350P,	Installation of HID Xenon LDG/TAXI and LED NAV Light	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	EASA STC 10038427 from NAYAK Aircraft Service GmbH based on: NAYAK PA46-NY-09199-MDL-to be operated in accordance with NAYAK AFMS PA46-NY-09199-08-to be maintained in accordance with NAYAK MM-P PA46-NY-09199-07 or later revisions of the above listed documents approved by EASA.	Limitations as referred to in AFMS. Must be maintained i.a.w. Original Piper MM, furthermore must be maintained i.a.w. Maintenance Instructions in the aforementioned Repair Instructions or later approved revisions. Prior to installation of this design change it must be determined that the interrelationship...	02/02/2018

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10039546	ROEDER PRAEZISION GMBH			TU206F	Installation of an additional Alternator with Battery for Mission Purposes	The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Flight Manual Supplement 01271AM240030-01A or later revisions of the above listed documents approved by EASA Instructions for Continued Airworthiness 01271CA240030-01A Engineering Order 01271EO240030-01A	None	10/05/2012
10039554	ROEDER PRAEZISION GMBH			172S	Installation of provision for the installation of a camera in the lower fuselage as described in description document 14576DE 5300 10 00D.	The Certification Basis of the original product remains applicable to this change. This certificate/ approval involves a change to the certificated noise and/ or emissions levels.	Master record list Rev A C172S 14576 MR 5300 10 01C (rev.C) dated 10.05.2012. EASA approved Supplement to Pilot's Operating Handbook 14576 AM 5300 1001C (rev.C) dated 27.04.2012 and any later EASA approved revision. Instruction for Continued Airworthiness 14576 CA 5300 10 01B (rev.B) dated 10.04.2012.	Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). This STC is limited to S/N 172S8186. This STC provides instructions only for the installation of the provision for a camera. Further instructions for the completion of the...	10/05/2012
10039755 (EASA.A.S.00504)	ROEDER PRAEZISION GMBH	REV. 2		A330-322	Cabin reconfiguration from a 387 Y/C layout to a 18 B/C + 331 Y/Clayout. This STC installs a new cabin arrangement with in-seat video and additional galley installations.	The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. As per EASA TCDSA.004. Elect to Comply with CS25 (Original Issue) requirements: 25.811 (d) / 25.812 (d) (e) / 25.813 (a) (b) (e) / 25.851(...	Master Document List, ref. MDL330-25-01-04 Revision 1, dated 09 Dec. 2004 (>,<) or later revisions of the above listed documents approved by EASA.	Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/06/2018

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10043115 (EASA.A.S.02769)	ROEDER PRAEZISION GMBH	REV. 1		737-800/-900 737-400/-500/-600/-700 737-100/-200/-200C/-300	737-DMD34-0 1-2006: Activation of Enhanced Surveillance Parameter.	The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Design Modification Document 737-DMD34-01-2006 Rev IR dated. 27.11.2006;-AFM supplement 737-AFM34-01-2006 Rev. IR dated. 27.11.2006; or later revision(s) of the above listed documents approved by EASA.	- Compliance to NP A 20-12a was shown.- Applicable to 737 models as per Design Modification Document 737-DMD34-0 1-2006.- This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation".C...	15/06/2018
10043852	ROEDER PRAEZISION GMBH	REV. 1		114, 114A, 114B, 114TC 112, 112B, 112TC, 112TCA	Installation of HID Xenon LDG/TAXI and LED NAV Lights. Replacement of existing installation with a new HID and LED installation.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Master Document List, Nayak GmbH, document RC114-NY-10202, rev. 1, dated January 2, 2013; Instructions for Continued Airworthiness and Installation according to Nayak GmbH, document RC114-NY-10202, rev. 0, dated October 26, 2011; Aircraft Flight Manual Supplement, Nayak GmbH, document RC114-NY-10202-08, o...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10045229	ROEDER PRAEZISION GMBH	REV. 1		PA-28RT-201T TURBO ARROW IV PA-28RT-201 ARROW IV PA-28R-201T TURBO ARROW III PA-28R-201 ARROW III PA-28R-200 ARROW II PA-28R-200 ARROW PA-28R-180 ARROW PA-28-236 DAKOTA PA-28-235 CHEROKEE PATHFINDER PA-28-181 ARCHER III PA-28-181 ARCHER II PA-28-180 CHEROKEE PA-28-180 ARCHER PA-28-161 WARRIOR III PA-28-161 WARRIOR II PA-28-160 CHEROKEE PA-28-151 CHEROKEE WARRIOR PA-28-140 CHEROKEE CRUISER	Installation of J.P. Instruments (JPI) Primary Engine Data Management System EDM930 in accordance with Engineering Order EDM930-NY03218-05. The EDM-930 is a combined electronic indicating system which simultaneously displays to the pilot powerplant and aircraft systems operating parameters.	The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23, Amendment 2 for affected paragraphs: 23.21, 23.303, 23.561, 23.603, 23.773, 23.1301, 23.1305, 23.1309, 23.1321, 23.1327, 23.1337, 23.1357, 23.1365, 23.1431, 23.1529, 2...	Airplane Flight Manual Supplement EDM930-NY-03218-08, Revision 0, 10.04.2012 or later revisions of the above listed documents approved by EASA Master Document List, Revision 0, 11.04.2012, or later approved revisions Installation Instruction: Engineering Order EDM930-NY-03218-05, Revision 0, 29.03.2012, a...	None	15/11/2017
10048031 (2004-5755 / LBA SA1390)	ROEDER PRAEZISION GMBH	REV. 2		SA227-TT, SA226-AT, SA227-AT SA227-AC, SA227-DC, SA227-BC SA226-T(B), SA226-T, SA226-TC	Installation of TCAS II System and Mode-S Transponder as per LBA STC No. SA 1390 Issue 1	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	EASA STC 2004-5755 from Nayak Aircraft Service GmbH based on: LBA STC No. SA1390 Issue 1, dated 04th June 2004.-AFM Supp. CAS 67 TCAS System and MST 67A Mode S System for Fairchild, Doc No. SA226/227-NY-1002-08, Rev 0, dated 21st May 2004. or later revisions of the above listed documents approved by E...	As listed in the AFM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	29/09/2017

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10049626	ROEDER PRAEZISION GMBH	REV. 1		A320-232	Galley G5 and Lavatory D installation. (Removal of galley G4, lavatory F, partition L4 and cabin attendant seat. Installation of new/refurbished galley G5, lavatory D and 2 cabin attendant seats. Modification of complete aft galley / entry area to adapt for these installations. Modification of structura...	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	-Master Document List, AirBerlin Technik document No. A320-MDL25-06-2014, Revision 4, 23 June 2014; -Instructions for Continued Airworthiness, AirBerlin Technik document No. A320-ICA25-06-2014, Revision 1, 20 June 2014; or later revisions of the above listed documents approved by EASA.	Limited to MSN 2587.	15/06/2018
10051268	ROEDER PRAEZISION GMBH	REV. 1		A320-214	Installation and activation of an In-Seat Power System on existing business class seats.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Master Document List A320-MDL25-15-2014 Rev. 01, dated 18/11/2014; Certification Report A320-CR25-15-2014 Rev. 04, dated 18/11/2014; Instructions for Continued Airworthiness A320-ICA25-15-2014 Rev. Initial, dated 11/11/2014; Aircraft Flight Manual Supplement A320-AFS25-15-2014 Rev. 01, dated 29/09/2014; o...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/06/2018

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10051425	ROEDER PRAEZISION GMBH	REV. 1		P.68 "VICTOR" P.68R "VICTOR" P.68C-TC P.68C P.68B "VICTOR" AP68TP-600 "VIATOR" AP68TP-300 "SPARTACUS"	Installation of PFD Aspen EFD1000 PRO and MFD Avidyne EX600with SatcomTransceiver MLX770 and TAS605	The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 3 (not required by 21A.101) 21, 29, 303, 305, 307, 561, 603, 609, 611, 613, 625, 771, 773, 867, 1301, 1303, 1309 (>,<)>1311, 1321, 1322, 1327, 1331, 1351 (>,<)>1353, 1357, 1359, 1...	Supplement to AFM: P68-NY-04249-08 Rev. 00 ,,, or later EASA approved revisions Compliance Program: P68-NY-04249-01 Rev.00 or later approved revisions,, Compliance Checklist: P68-NY-04249-01a Rev. 00 or later approved revisions,, Groundtest Requirements: P68-NY-04249-04a Rev. 00 or later approved revisi...	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017
10053859	ROEDER PRAEZISION GMBH	REV. 1		A320-214, A320-216	Activation of PED power system on Recaro 3520D seats.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Certification Report A320-CR25-03-2015 Rev. 01, dated 26/06/2015 Master Documentation List A320-MDL25-03-2015 Rev. 00, dated 26/06/2015; Instructions for Continued Airworthiness A320-ICA25-03-2015 Rev. 01, dated 26/06/2015; Aircraft Flight Manual Supplement A320-AFS25-03-2015 Rev. 00, dated 27/05/2015; or...	None	15/06/2018
10056736	ROEDER PRAEZISION GMBH	REV. 1		EMB-505	Installation of Low Gain Satcom Antenna	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Master Document List No. LGA-NY-11277 Rev. 00 Documentation ICA Instruction for Continued Airworthiness No. LGA-NY-11277-07 Rev. 01 or later revisions of the above listed documents approved by EASA.	Airworthiness limitations resulting from this modification according to Document: ICA Instruction for Continued Airworthiness No. LGA-NY-11277-07 Rev. 01 (Chapter 4.1); (Threshold- and repeat Inspection Interval for aircraft skin and doubler). Prior to installation of this design change must be determined t...	15/11/2017



EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10061997 (2004-12063 // LBA EMZ SA 1412)	ROEDER PRAEZISIO N GMBH			M20, M20A, M20B, M20C, M20D M20L, M20SM, M20R, M20S M20E, M20F, M20G, M20J, M20K	Installation of Bendix King Mode S Transponder KT 73	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	EASA STC 2004-12063 from Nayak Aircraft Service GmbH based on: LBA EMZ SA 1412 Issue 1, dated 10th December 2004.-AFM Supp. Doc. : M20-NY-0706-08, Rev 0 or later revisions of the above listed documents approved by EASA.	As listed in the AFM Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	12/10/2017
10063239 (EASA.A.S.00332)	ROEDER PRAEZISIO N GMBH	REV. 1		SA227-DC, SA227-TT, SA226-AT SA227-AT, SA226-T(B), SA226-T, SA226-TC, SA227-AC, SA227-BC,	Installation of MST 67 Mode-S Transponder.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	AFM Supplement SA226/227-NY-0303-08, Rev. 0, or later EASA approved revision.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	25/09/2017
10063240 (EASA.A.S.00639)	ROEDER PRAEZISIO N GMBH	REV. 1		SA227-DC, SA227-TT, SA226-AT SA227-AT, SA226-T(B), SA226-T, SA227-AC, SA227-BC,	Installation of a Garmin GTX 330D Mode S Transponder.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- Flight Manual Supplement Doc. No. SA226/227-NY-0604-08 or later approved revision.- MM Supplement SA226/227 NY-0604-07 or later approved revision.	Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	25/09/2017
10063400 (2004-11566 // LBA EMZ SA 1417)	ROEDER PRAEZISIO N GMBH			114, 114A, 114B, 114TC 112, 112TC	Installation of Bendix King Mode S Transponder KT 73	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	EASA STC 2004-11566 from Nayak Aircraft Service GmbH based on: LBA EMZ SA 1417 Issue 1, dated 30th November 2004.-AFM Supp. Doc. : RC 114-NY-0707-08, Rev 0 or later revisions of the above listed documents approved by EASA.	As listed in the AFM Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	16/10/2017

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10063401 (2004-12063)	ROEDER PRAEZISION GMBH			M20, M20A, M20B, M20C, M20D, M20L, M20S, M20M, M20R M20E, M20F, M20G, M20J, M20K,	Installation of Bendix King Mode S Transponder KT 73	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Approval as per EASA STC No. 2004-12063. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applic...	EASA STC 2004-12063 from Nayak Aircraft Service GmbH based on:- LBA EMZ SA 1412 Issue 1, dated 10th December 2004.- AFM Supp. Doc. : M20-NY-0706-08, Rev 0. or later revisions of the above listed documents approved by EASA.	As listed in the AFM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017
10063402 (2004-5755)	ROEDER PRAEZISION GMBH			SA227-TT, SA226-AT, SA227-AT SA227-AC, SA227-DC, SA227-BC, SA226-T(B), SA226-T, SA226-TC,	Installation of TCAS II System and Mode-S Transponder as per LBA STC No. SA 1390 Issue 1	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Approval as per EASA STC No. 2004-5755. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applica...	EASA STC 2004-5755 from Nayak Aircraft Service GmbH based on:- LBA STC No. SA1390 Issue 1, dated 04th June 2004.- AFM Supp. CAS 67 TCAS System and MST 67A Mode S System for Fairchild, Doc No. SA226/227-NY-1002-08, Rev 0, dated 21st May 2004. or later revisions of the above listed documents approved by...	As listed in the AFM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017
10063403 (2004-11566)	ROEDER PRAEZISION GMBH			114TC 114, 114A, 114B, 112, 112TC,	Installation of Bendix King Mode S Transponder KT 73	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Approval as per EASA STC No. 2004-5755. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applica...	EASA STC 2004-11566 from Nayak Aircraft Service GmbH based on:- LBA EMZ SA 1417 Issue 1, dated 30th November 2004.- AFM Supp. Doc. : RC 114-NY-0707-08, Rev 0. or later revisions of the above listed documents approved by EASA.	As listed in the AFM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.	15/11/2017

EASA STC no.	STC Holder	Rev. level	Foreign/NAA STC ref.	Models	Description	Certification basis	Limitations	Conditions	Date of Issue
10063409 (EASA.A.C.12747)	ROEDER PRAEZISIO N GMBH			M20/M20A/M20B/M20C/M20D M20L/M20M/M20R/M20S M20E/M20F/M20G/M20J/ M20K	Installation of GNS 430 WAAS navigation system.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	Master Document List Doc. No. GNS430-NY-05131-EB06. Engineering Order No. GNS430-NY-05131-05. AFM-S Document No. GNS430-NY-05131-08, Rev 0. ICA Document No. GNS430-NY-05131-07, Rev 0 or later revisions of the above listed documents approved by EASA.	As stated in limitation section NAYAK AFM-S Document No. GNS430-NY-05131-08, Rev 0. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon...	15/11/2017
EASA.A.S.03633	ROEDER PRAEZISIO N GMBH			CIRRUS SR22 CIRRUS SR20	Installation of Avidyne Iridium MLX770 Two Way Datalink.		- According to Röder Präzision MRL Document No. MR 3400-46-01A, Rev. 3 or later Revisions- To be operated in accordance with approved Röder Präzision AFM-S Document No. 2882AF 3400-46-01A, Original Issue, or later approved Revisions, or- To be operated in accordance with approved Röder Prä...	1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by thi...	03/12/2008
10015256 (EASA.A.S.03709)	ROEDER- PRAEZISIO N GMBH	REV. 1		SA226-T(B)	Installation of SLAR Surveillance System quick change roll.	The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.	- EASA approved „Übersicht der Dokumente“, Project No. SA226-NY-10117, Revision 1, dated 19 February 2009, or later approved revision.- EASA approved Airplane Flight Manual Supplement, Doc. No. SA226-NY-10117-08, Revision 2, dated 22 December 2008, or later approved revision.- Maintenance Procedures, 0...	1. Limited to Serial Number T294. 2. With SLAR equipment installed, maximum cabin seating is 4 + 1. Operator Seat must not be occupied during take-off and landing. 3. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other prev...	30/10/2017